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ON THE INTERPOLATIVE ANOMALIES FOR THE FIRST TEN MINOR PLANETS

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ON THE INTERPOLATIVE ANOMALIES FOR THE FIRST

TEN MINOR PLANETS

(Ob interpolyatsionnykh anomaliyakh dlya pervykh desyati malykh planet)

(USSR)

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by M. YAROV - YAROVOY

The aim of the present paper is the study of the question of the so-called "interpolative anomaly" for minor planets.

The notion of interpolative anomaly has been introduced by N. D. Moiseyev in his work under reference [1], devoted to the description of a new type of averaged variants of the restricted three body problem. This new type was designated by Moiseyev as the "interpolatively-averaged variant" of the restricted three body problem. It has a well known analogy with the Delaunay-Hill averaged problem, of which it is a substantial generalization.

The main feature of this generalization is the substitution of the so-called Delaunay anomaly by a linear combination of angular expansion variables of a perturbation function selected on the

^{*} Trudy Gosudarstvennogo Astronomicheskogo Instituta imeni P.K. Shternberga (Transactions of the P.K. Shternberg State Astronomical Institute).

basis of interpolative processing of the observation material, or of the examined minor planet's or comet's osculating elements, corresponding to various moments of some examined time interval. At the same time, this processing is channelled toward searching for a linear combination of the mentioned angles, that would remain "almost constant" over the examined time interval. In the present work we are attempting to establish an interpolative anomaly for the first ten minor planets.

In this context, the present paper may be considered as the first experiment in investigating the possibility of establishing an "interpolative anomaly" without which it is difficult to have an opinion on the practical applicability of the above-mentioned "interpolalatively-averaged schemes" of the restricted three body problem, introduced by Moiseyev in the work, to which reference has been made above.

We shall utilize in this work the scheme of the restricted circular problem of three bodies: Sun, Jupiter, Asteroid, where the Sun is the perturbing body, moving along a mean circular orbit with a radius equal to the great semi-axis, and with an inclination and longitude of the orbit plane's node equal to those of the mean elliptical orbit of Jupiter for the year 1850, computed by Hill. The Asteroid is the perturbed body.

#1. UTILIZED OSCULATING ELEMENTS' SYSTEMS OF THE FIRST TEN MINOR PLANETS

The systems of osculating elements for the first ten minor planets, utilized for the computation of interpolative anomalies, are presented below. (See Tables in the next and the following pages).

Table 1

Angular Elements of the Orbits of the First Ten Asteroids *

	85.07.1.0 5.07.1.0 6.00 7.00 7.00 8.00 8.00 8.00 8.00 8.00 8		259.7 28.4 28.5 2.55.5 32.6 32.6 32.6
(e)	27.28 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8		88668686448944
	000000000000000000000000000000000000000		*************
	30.5.2 30.5.2 30.5.3 30.5.3 30.5.3 30.5.3		20.24 20.24 20.24 37.44 37.44 20.27 20.27 20.27 20.27
(a) (S	80° 58° 58° 58° 58° 58° 58° 58° 58° 58° 58		22.122.288.888.888.1888.1888.1888.1888.
(735, 4 18 28 38 32.6 32.6 32.5 32.5 12.6 11.0		717 25.1 35.1 22.3 30.3 30.3 44.3 31.6 31.3
(a) ^(a)	64° 58′ 65° 65° 78° 65° 78° 65° 78° 79° 79° 79° 79° 79° 79° 79° 79° 79° 79	5.	309° 12 308° 308° 308° 308° 308° 308° 308° 308°
	ERES	54776	
,	7	. PA	**************************************
W	76. 7 290° 30 2290° 45 2290° 45 2290° 45 1113 22 113 22 113 22 113 22 113 22 113 22 113 22 113 22 113 22 110 20 270.032 256.080 11.511	No 2.	41° 16′ 100 26 133 48 133 48 133 28 148 5 267 46 261.594* 110 111
٠			
Eclipt	1801 1801 1801 1818 1818 Ep. Ep. 1925 1925 1950 1950		Ep. 1803 1803 1803 1810 Ep. Ep. Ep. 1925 1925
		_'	
och	15.0 0.0 0.0 0.0 0.0 0.0 5.0 7.0 7.0	1	31.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0
Oscul.epoch	Jan. Jan. Jan. Jan. Jan. Jan. Jan. Jan.	1	ପ୍ରଷ୍ଟ୍ର ପ୍ରଦ୍ରକ୍ଷ ଓଡ଼ା
Osc	1801 1850 1854 1854 1951 1951 1952 1938 1938	 	1802 1803 1803 1810 1810 1810 1810 1913 1913 1914
No.	- 2 884886888831384		<u> </u>
_			

st The references ahead of the second column bear on the literature sources from which they are taken

** M is given on 1.0.1. 1925.

21		·			
169	-d		34.736 11 46 33.9 34.736 34.804 34.818 34.825 34.818		13° 4'26".2 13 4 12.9 13 4 119.0 13 4 17.2 13 3 47.3 13 3 47.3 13 0.3 13 0.3 13 0.3 13 0.3 13 0.0 13
Sle			172.962 159.56 172.897 173.069 173.039 173.033		171° 4'28".2 171 6 45 .0 171 6 28 .5 171 9 16 .7 171 9 58 .9 171 8 11 .1 170 59 49 .7 170 59 49 .7 170 30 12 .7 170 30 12 .7 170 430 11 20 20 .2 170 430 170 437 170 437 170 374
	(a) (a)	continuation).	309.695 323 24 37.5 309.281 309.850 310.142 310.275		242° 14'32".0 241 59 58.0 241 52 74.4 242 5 53.4 242 22 55.4 243 22 55.4 243 9 18.1 244.9 245.603 245.603 245.603 245.603 245.603 245.603
	М	PALLAS (C	259.753* 323 49 13.7. 260.419* 245.220 348.968 284.962 301.274	. EUNO	349° 187° 37.5 42° 283 10° 2 42° 283 10° 2 124 33 10° 3 176 58 40° 4 176 38 15° 5 351 4 24 57 7 4 24 58 0 317° 57 26 6 317° 57 26 6 317° 26 8 172° 26 8 172° 26 8 172° 26 8 170° 38 8 299° 895
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	No.		45.25.25.25.25.25.25.25.25.25.25.25.25.25		<u> </u>

Table 1 (cont. 3)

(e)	13°.017 13 .013 12 .993		7. 8,117.6 7. 8 5.0 7. 8 9.8 7. 8 9.8 7. 8 6.6 7. 131			55 19 25 19	247, 1 44, 1 46, 47, 1 735, 3 75, 1 75, 1	
S)(e)	t'n - 170°.711 170 .702 170 .496		103° 8′20″.5 103 11 22 .1 103 13 29 .0 103 11 38 103 23 19 .1 104 596	.007 26 23 .3 9 53 .2		141°26′16″.6 141°25′46.76 141°25°14°.6 141°25°13°.6 141°25°8°.9 141°24°48°.5	က်ကလူလု	
(e)	ение) Е и NO - сот 245°.378 245°.477 246°.323		146° 40′ 6″.4 146° 8° 6° .6 146° 29° 6° .3 146° 20° 6 147° 10° 58° .7 147° 10° 40° .2 147° 10° 40° .2		والمساول والمتاركة والمتار	353°58′ 5″.7 353 52 34 .3 353 55 32 .4 353 55 44 .9 353 11 14 .8 353 10 47 .2 353 59 42 .3	్రాఖంచినిటెన్ లా బైజ్ఞ4్ల	
W	Ю но на (продолжение) 62°.905 164.041 248.554	YESTA.	216° 4'48".7 216 42 55 .8 91 25 8 .0 290 6 46 198 20 17 .2 198 20 32 .8 198.208	``	ASTREA.	318° 40' 43" 7 318 51 49 .0 318 45 3 .3 318 45 2 .7 306 19 34 .1 306 20 27 .0		101,000
Eclip.	№ 3. К 1950 1950 1950	No A	1810 1810 Ep. 1818 Ep. Ep.	1950 Ep. 1925e	No 5.	Ep. 1846 1846 1846 1850 Ep. Ep.	1910 1947 1847 1870 1900 1950	OCRI
Oscu l. Epoch	F E B 1941 февр. 15.0 MAY 1942 Mail 9.0 Aug 1948 abr. 7.0		1810 янв. 1810 янв. 1816 янв. 1816.0 1857 янв. 1857 янв.	747 1857 янв. 2.0 744У 1869 май 13.0 747. 1925 янв. 20.0		1846 янв. 1846 янв. 1846 янв. 1846 янв. 1850 янв. 1850 янв.	1898 1900 1847 1870 1900	февр
No.	205)		<u> </u>	<u> </u>		566666	SS 5584 F	<u>_</u>

Table 1 (cont. 4)

(e)		14°.752		107.2 53.51 55.8 55.8 55.8 75.28 17.1 44.3 55.8 75.485 75.485 75.488 75.498 75.498 75.498 75.498 75.498 75.498 75.498 75.498		12".0 5°53'4".8 12 .8 5 53 5.4 3 .8 5 53 6.2
(a)(c)		138°.911 138.883 138.858		259 48' 259 44' 259 475 259 475 250 83 250 827 250 674 250 676 250 676 250 676 250 676 250 676 250 676 250 676 250 676 250 676 250 676 250 676 250 676		110°18' 110 18 110 18
(e)	(cont'n).	238°.113 238.189 238.192		141° 53′ 3″.3 141° 40 16.8 141° 35 18 6.8 141° 35 25 .3 141° 490 143.238 143.238 143.902 143.902 143.819 143.715 143.715	,RA	282°42'28".8 282°43'3 3.2 282°42 19.4
M	OG. HEBE	215°.829 351.352 187.138	No 7 1Ris	330° 41' 54".0 330° 56 26° .6 166° 7° 39° .3 166° 7° 9° .0 9° 520° .1 280.717* 280.288* 289.984* 289.988* 289.918* 289.101* 289.101* 289.101* 289.101* 289.101* 289.101* 289.101*	No 8 FLORA	35°48' 7".0 35 47 35 .1 35 48 24 .3
Eclip.	No	1950 1950 1950		EP. 1850 1850 1850 1925 1925 1925 1950 1950		1848 1848 1848
Oscul. Epoch		74 y 1941 mañ 6.0 0c7 1942 okt. 8.0 4 LG 1948 abr. 7.0		7AN 1848 янв. 1.0 7AN 1848 янв. 1.0 7AN 1850 янв. 0.0 7AN 1850 янв. 0.0 7AN 1950 янв. 0.0 6CT 1931 окт. 1.0 8EP 1933 сент. 15.0 8EP 1934 сент. 15.0 6CT 1935 окт. 1.0 6CT 1936 окт. 1.0 6CT 1936 окт. 1.0 6CT 1938 июль 4.0 7UNE 1939 июнь 30.0 7UNE 1940 июнь 24.0 7UNE 1941 июнь 19.0	p	74 № 1848 янв. 1.0 2.4 № 1848 янв. 1.0 2.6 № 1848 янв. 1.0
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Table 1 (cont.5)

		8 27 46".8 5 35' 48".6 8 27 46 .7 5 35 48 .5 8 29 30 .5 5 5 35 0 .3 9.686	27, 46".8 5° 35' 48" 27 46".7 5° 35 48 27 50 .1 5° 35 48 29 30 .5 5° 35 57 31 35 .2 5° 35 0 686 5° 603
		33.24".3 68° 27' 33 42 .3 68 27 33 39 .3 68 27 42 15 .0 68 27 656 69.686	247.3 68° 27 442.3 68° 27 39.3 68 27 15.0 68 21 16.9 68 31
_		2 33724" 2 33 424" 2 42 15 2 22 16	68824888
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_		744 7444 7444 7676 7676 7676	748 748 7488 5086 5086
		<u> </u>	\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$

Here the first column indicates the system number of the osculating elements for every asteroid; the second column shows the osculation epoch. The third column gives the epoch of the equinox, to whose ecliptic the elements ω , Ω and i are related. The sign "e", whenever it stands near that epoch, indicates that the corresponding elements are related to the equator. Further we have: M, $\omega^{(e)}$, $\Omega^{(e)}$ and i (e). The upper index "e" at the three latter indicates that they are either related to the equator or to the ecliptic. But what is lacking in the preceding Table, it is those osculating element systems that are clearly erroneous as a result of computations, though given in literature, of ω , Ω and i that have been related to ecliptics of different equinoxes.

2. UTILIZED ELEMENT SYSTEM OF THE MEAN ELLIPTICAL ORBIT OF JUPITER

This system of elements, computed by Hill for the year 1850, is presented below:

$$\lambda = 159^{\circ}56'25''.05,
\pi = 11 54 26 .72,
\Omega = 98 55 58 .16,
i = 1 18 41 .81.$$

$$e = 0.04825382,
n = 299''.12837656,
lg a = 0.7162373716.$$

The system of elements of Jupiter's orbit was borrowed by us from the "Astronomical Yearbook". The elements ω , Ω , and i were calculated over again relative to the ecliptic of the year 1950.0 equinox to be used as a basis for further computations:

$$\begin{cases}
\Omega = 99^{\circ}46'42''.6 \\
\omega = 273 \ 31 \ 28 \ .0 \\
i = 1 \ 18 \ 29 \ .2
\end{cases}$$
1950.0

#3. ON THE COMPUTATION OF AUXILIARY QUANTITIES

In order to interpolate the values of the anomalies, it is necessary to know those of the mean anomaly for the various moments of time; of the angular distance of the perihelion from the node, specifically — from the node of the plane of the asteroid's orbit in the orbital plane of Jupiter, and not in the plane of the ecliptic; of the node's longitude computed from the Jupiter's radius-vector in the latter's orbital plane to the line of nodes of the orbital plane of the asteroid in the orbital plane of Jupiter. Since all the three quantities are not available in the Table brought out in #1, we must proceed with their computation.

The first quantity— the mean anomaly of the asteroid M — is obtained by accounting for the number of revolutions made by the minor planet from the moment of osculation for the first system of its elements, and by corresponding additions to the values M of multiples of 360°.

As to the remaining two quantities — the angular distance of the perihelion from the node ω and the longitude of the node Ω — the composition of a special computing scheme for them was made necessary, for these quantities ω , Ω are given in the table of elements of minor planets relative to the ecliptic plane, and not to that of Jupiter's orbit, and for the ecliptic of various equinoxes. Thus emerges the problem of translating these elements from the ecliptic planes of different epochs to the plane of Jupiter's mean circular orbit.

This problem is solved in two stages: first of all one must compute these elements relative to the ecliptic of any one epoch, for example as is now the case, relative to the ecliptic of the year 1950, then transfer these elements from the ecliptic plane of the year 1950 to the plane of Jupiter's mean circular orbit. In conformity with the computing schemes, such an operation, as we shall see in a moment, must not only be conducted with elements ω and Ω , but also with the orbit inclination i of the asteroid.

The first half of the problem is solved with the aid of the following formulae (see for instance, the book by A. Ya. Orlova and B. A. Orlova [2]):

$$\Omega_{1950}^{(e)} = \Omega_{t}^{(e)} + a - b \sin \left(\Omega_{t}^{(e)} + c \right) \operatorname{ctg} i_{t}^{(e)},
i_{1950}^{(e)} = i_{t}^{(e)} + b \cos \left(\Omega_{t}^{(e)} + c \right),
\omega_{1950}^{(e)} = \omega_{t}^{(e)} + b \sin \left(\Omega_{t}^{(e)} + c \right),$$
(1)

where the values of the coefficients <u>a</u> and <u>b</u>, and also of the quantity <u>c</u>, are given for different equinoxes in Table VII, at the end of the book referred-to [2].

With the help of the obtained values of $\omega_{1950}^{(e)}$, $\Omega_{1950}^{(e)}$ $\ll i_{1950}^{(e)}$ one may also resolve the second part of the problem. We shall examine for that purpose the spherical triangle formed on the celestial sphere by the lines of intersection with it of the ecliptic planes, of Jupiter's mean circular orbit and of the osculating orbit of the asteroid. (Fig. 8).

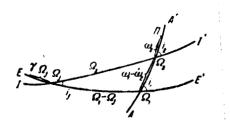


Fig. 8

EE' — Position of the ecliptic in the year 1950; Ω — point of the spring equinox; II' — position of Jupiter's mean orbital plane; Ω_j — Node of Jupiter's mean orbit; i_j —its inclination; AA' — Position of the osculating plane of the asteroid; π —Perihelion of the orbit of the asteroid; Ω_j —asteroid orbit's node situated in the ecliptic; Ω_j —asteroid orbit's node situated in the mean orbital plane of Jupiter; Ω_j , i_j ,

From the spherical triangle ∂_1 , ∂_2 , ∂_j one may determine the quantities ∂_2 , $\omega_1 - \omega_2$ and i_2 by means of the following spherical trigonometry formulae:

$$\operatorname{tg} \Omega_{2} = \frac{\sin \left(\Omega_{1} - \Omega_{j}\right)}{-\operatorname{ctg} i_{1} \sin i_{j} + \cos i_{j} \cos \left(\Omega_{1} - \Omega_{j}\right)}, \\
\operatorname{tg} i_{2} = \frac{\sin \left(\Omega_{1} - \Omega_{j}\right)}{\sin \Omega_{2} \left[\operatorname{ctg} i_{1} \cos i_{j} + \sin i_{j} \cos \left(\Omega_{1} - \Omega_{j}\right)\right]}, \\
\operatorname{tg} \left(\omega_{1} - \omega_{2}\right) = \frac{\sin \left(\Omega_{1} - \Omega_{j}\right)}{\sin i_{1} \operatorname{ctg} i_{j} - \cos i_{1} \cos \left(\Omega_{1} - \Omega_{j}\right)}.$$
(2)

This provides the possibility of computing the values ω , i and $\overline{\mathcal{X}}$ searched for. The latter is obtained from the formula $\overline{\mathcal{X}} = \mathcal{X}_2 - \mathbf{u}_j$ where \mathbf{u}_j is the argument of Jupiter's latitude (which we assume to be moving along a circle).

#4. UTILIZED SYSTEMS OF OSCULATING ANGULAR ELEMENTS OF THE FIRST TEN MINOR PLANETS BROUGHT TO THE SAME FORM

The thus carried out computations gave the following elements:

Table 2

Ne π/π	м	œ	શ	· i
		№ 1. Цере		,
1 2 3 4 5 6 7 8 9 10 11 12 13 14	291°.506 290 .766 1 680 .327 4 121 .033 4 433 .381 4 433 .306 4 433 .370 9 073 .886 9 670 .340 10 056 .684 11 070 .032 11 156 .080 11 171 .511 11 830 .382	62°.761 63 .420 64 .950 65 .324 65 .723 65 .737 65 .727 66 .149 69 .208 68 .918 69 .647 68 .511 68 .500 67 .239	+ 1406°.408 1406 .339 + 866 .121 - 81 .499 202 .906 202 .912 2004 .923 2237 .830 2387 .270 2780 .080 2814 .347 2820 .333 - 3075 .524	9°.389 9.389 9.380 9.381 9.378 9.378 9.369 9.369 9.367 9.367 9.368
			ада - (РАЦАS)	
1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20	41°.274 100 .078 100 .446 133 .612 646 .250 647 .854 1 933 .391 2 208 .096 5 307 .777 8 711 .659 8 977 .583 9 110 .188 9 418 .704 9 619 .753 9 683 .821 10 222 .360 10 685 .220 10 788 .968 11 084 .962 11 461 .274	306°.961 306 .696 306 .425 306 .416 306 .364 306 .109 306 .113 306 .747 306 .788 307 .791 307 .426 307 .426 307 .426 307 .423 307 .426 307 .635 307 .927 308 .061 307 .793	+ 1464°.292 1441 .451 1441 .454 1428 .487 1228 .952 1228 .980 728 .685 + 621 .830 - 584 .539 1908 .366 2011 .816 2063 .932 2184 .206 2262 .320 2287 .227 2496 .661 2677 .235 2717 .698 2832 .731 - 2978 .813	34°.256 34.332 34.281 34.285 34.260 34.305 34.252 34.357 34.348 34.348 34.381 34.388 34.382 34.380 34.381 34.447 34.464 34.469 34.461
. 1	349°.301	№ 3. Юно: 236°.552	на - (Euno) + 1382°.988	12°.740
1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22	762 .386 762 .739 844 .553 1 256 .978 1 504 .202 1 616 .638 2 151 .724 5 044 .416 8 188 .600 9 317 .957 9 570 .988 10 066 .444 10 382 .951 10 808 .402 11 042 .130 11 357 .958 11 459 .895 11 582 .905 11 684 .041 12 198 .554	236 .310 236 .408 236 .408 236 .391 237 .748 236 .692 236 .546 237 .477 239 .234 240 .041 239 .223 239 .923 239 .946 239 .949 240 .177 239 .976 239 .833 239 .723 239 .723 239 .723 239 .723	+ 1382 : 398 1231 : 232 1231 : 220 1200 : 967 1049 : 150 957 : 984 916 : 694 + 720 : 143 - 344 : 444 344 : 447 1501 : 831 1917 : 762 2010 : 171 2192 : 884 2309 : 316 2465 : 943 2552 : 002 2667 : 999 2705 : 393 2750 : 601 2787 : 697	12 .746 12 .739 12 .739 12 .733 12 .734 12 .728 12 .705 12 .704 12 .671 12 .632 12 .644 12 .641 12 .649 12 .651 12 .652 12 .649 12 .651 12 .652 12 .646 12 .646 12 .646 12 .646

Table 2 (contin'd)

№ п/п	М	ω,	হ	
		№ 4. Bec	ta (Vesta)	
1 2 3 4 5 6 7 8 9	216°.080 216 '716 811 '.419 1010 .113 4878 .338 4878 .342 4876 .339 4878 .208 6105 .231 11627 .508	145°.646 145 .099 145 .453 145 .314 146 .194 146 .189 147 .326 146 .222 145 .671 147 .309	+ 1158°.718 1158 .791 976 .688 + 915 .878 - 268 .137 268 .137 268 .857 268 .136 643 .510 - 2334 .112	5°.829 5 .828 5 .829 5 .822 5 .822 5 .831 5 .861 5 .824 5 .831
		№ 5. Аст	рея (Astrca)	
1 2 3 4 5 6 7 8 9	318°.679 318 .864 318 .751 318 .751 666 .326 666 .341 1807 .014 4904 .067 5017 .580	342°.446 342 .333 342 .410 342 .413 341 .688 341 .681 342 .492 342 .070 342 .830	+ 114°.424 114 .395 114 .401 + 114 .401 - 7 .070 7 .067 404 .957 1485 .630 - 1524 .926	4°.444 4 .445 4 .445 4 .445 4 .448 4 .438 4 .439 4 .456 4 .451
•		№ 6. Геб	sa (Hebe)	
1 2 3 4 5 6 7 8 9	275°.225 275 .478 2498 .047 5324 .339 8323 .973 9101 .095 9215 .829 9315 .352 9907 .138	233°.807 233 .817 233 .209 233 .462 233 .891 234 .559 234 .641 234 .719 234 .724	+ 57°.194 + 57 .193 - 650 .988 1551 .313 2506 .655 2754 .640 2791 .196 2834 .435 - 3011 .447	13°.815 13 .814 13 .794 13 .812 13 .811 13 .764 13 .761 13 .762 13 .765
.45		№ 7. Ири	да (Jzis)	
1 2 3 4 5 6 7 8 9 10 11 12 13 14 15	330°.698 330 .941 526 .128 526 .119 5049 .089 6685 .222 8148 .999 8340 .072 8437 .115 8538 .591 8637 .001 8807 .926 8904 .537 9000 .941 9067 .263	138°.159 137 .936 137 .857 137 :859 137 .861 137 .553 139 .411 139 .529 139 .826 140 .058 140 .054 140 .003 139 .998 139 .894 139 .880	+ 164°.245 164 .200 103 .560 + 103 .562 - 1414 .855 1922 .205 2378 .256 2437 .712 2468 .127 2499 .701 2530 .278 2583 .466 2613 .464 2643 .318 - 2673 .317	6°.713 6.713 6.709 6.710 6.709 6.711 6.712 6.720 6.731 6.733 6.733 6.733 6.733 6.733

Table 2 (end)

№ п/п	М	w	શ	i i
		№ 8. Флој	pa (Flora)	· C
1 2 3 4	35°.837 5765 .680 7613 .404 9590 .690	279°.514 280 .684 279 .537 280 .663	+ 14°.186 - 1565 .397 2072 .304 - 2617 .771	4°.608 * 4.627 4.612 4.624
		№ 79. Мети	да (Metis)	
1 2 3 4 5 6	144°.328 145 .114 145 .114 1136 .944 1137 .076 1137 .218	11°.033 11 .038 11 .037 11 .186 11 .040 11 .977	- 49°.135 49 .384 49 .383 357 .855 357 .842 - 357 .894	4°.515 4 .515 4 .514 4 .518 4 .503 4 .519
		№ 10. Хиги	en - (Hygica)	-
1 2 3 4 5 6 7 8 9 10	481°.591 481 .532 3531 .338 4684 .331 4991 .678 4993 .536 5221 .962 5665 .099 6066 .697 6233 .586 6739 .084	306°.002 305 .814 310 .601 310 .626 306 .247 306 .242 306 .451 306 .112 306 .084 307 .558 311 .714	+ 73°.162 + 73 .495 - 1362 .647 1902 .242 2045 .672 2045 .752 2152 .985 2360 .763 2550 .329 2628 .960 - 2869 .710	5°.098 5 .092 5 .117 5 .117 5 .104 5 .112 5 .109 5 .102 5 .105 5 .77

^{*} The average was taken from five values of the systems of osculating elements for 1848. I. 1.0.

Given are in the preceding Table: number of the system of elements taken from #1; the mean anomaly M, computed by the above-described method for the osculation moment; the angular distance of the perihelion from the node of the plane of asteroid's orbit in the orbital plane of Jupiter (ω); the longitude of the node $\widehat{\lambda}$, computed by the just-described method, and the inclination of the plane of asteroid's orbit to the orbital plane of Jupiter.(i).

#5. GENERAL REMARKS ON THE METHOD OF COMPUTATION OF INTERPOLATIVE ANOMALIES

According to N. D. Moiseyev [1], the quantities M, ω , and $\overline{\Omega}$ are part of the expression for the interpolative anomaly μ as follows:

$$\mu = s_1 M + s_2 \omega + s_3 \Omega, \qquad (3)$$

where s_1 , s_2 and s_3 are some constant coefficients.

The problem of determining or establishing the interpolative anomaly μ consists in so selecting the coefficients s_1 , s_2 and s_3 , that the interpolative anomaly μ , determined by the formula (3), remain "almost constant" for the given asteroid during the examined time interval, encompassed by the utilized systems of osculating elements. Such problems requires either statistical processing of the system of osculating angular elements of the asteroid. One of such possible methods is the standard method of least squares. Here we compose a conditional equation of the form (3), where the coefficients s_1 , s_2 and s_3 , just as are considered unknown. After that a corresponding system of normal equations is being composed, from which the most probable values of s_1 , s_2 , s_3 and μ are derived.

The correlation method for numelous variables may serve as the second method for the solution of the same problem. Here formula (3) is considered as a regression equation with unknown coefficients s_1 , s_2 and s_3 . The equivalence of both methods is obvious. Indeed, in both cases considered is the minimum of certain sum S, which in our case is equal to

$$S = \sum_{g=1}^{n} (s_1 \delta M_g + s_2 \delta \omega_g + s_3 \delta \overline{\Omega}_g)^2,$$

$$\delta M_g = M_g - \overline{M}, \quad \delta \omega_g = \omega_g - \overline{\omega}, \quad \delta \overline{\Omega}_g = \overline{\Omega}_g - \overline{\overline{\Omega}};$$

$$(g = 1, 2, ..., n).$$
(5)

where

 \overline{M} , ω and \overline{M} are the mean values of our quantities M_g , ω_g and \overline{M}_g ; g is the number of their values borrowed from the initial Table.

Table of # 4 is in our case such a table for each of the examined ten asteroids.

Rather than apply in the present work one of the two above-indicated methods of determination of the quantities s_1 , s_2 , s_3 and μ , we shall make use here of a certain combination of both methods, leading to the objective the quickest possible way: In order to demonstrate the existence of a general and partial correlative link between the quantities M, ω , δ and to compute the coefficients s_1 , s_2 , s_3 themselves, we shall utilize the apparatus of the theory of linear correlation. As to the computation of errors in the coefficients s_1 , s_2 , s_3 and in the interpolative anomaly μ , both methods are used for that purpose — the method of the solution of normal equations, as well as the theory of linear correlation for many variables.

These methods are described in Romanovskiy's book [3] in the general case. We shall describe in the next paragraph the formulae and the method of computation applied in the current work.

#6. FORMULAE OF THE METHOD APPLIED FOR FINDING THE INTERPOLATIVE ANOMALY AND THE COEFFICIENTS s_1 , s_2 and s_3

In regard to coefficients s_1, s_2, s_3 and the quantity , it ought to be noted that they may be only obtained with a precision to a certain common numerical factor. Thus, for the purpose of definiteness, we shall alternately postulate the coefficients at M, ω , and $\overline{\Omega}$ as equal to the unity. This will liberate us from the just-indicated uncertainty.

For the sake of definiteness, let the coefficient at M by the unity. The cases of the coefficients equalling the unity at ω and $\bar{\lambda}$ are examined analogously.

Thus, we consider the interpolative anomaly of the form:

$$\mu_1 = M + s_2^{(1)} \mathbf{o} + s_3^{(1)} \overline{\Omega}. \tag{7}$$

We already mentioned, that for finding s_1, s_2, s_3 we must find the minimum of the sum:

$$S^{(1)} = \sum_{g=1}^{n} (\partial M_g + s_2^{(1)} \delta \omega_g + s_3^{(1)} \delta \overline{Q}_g)^2$$
 (8)

by a proper assortment of coefficients $s_2^{(1)}$ and $s_3^{(1)}$. In nonsingular cases the coefficients $s_2^{(1)}$ and $s_3^{(1)}$ are found from the equations:

$$\frac{\partial S^{(1)}}{\partial s_2^{(1)}} = 0; \quad \frac{\partial S^{(1)}}{\partial s_3^{(1)}} = 0. \tag{9}$$

Provided these equations are written in expanded form, we may substitute the quantities δM_g , $\delta \omega_g$ and $\delta \overline{\delta}_g$ by the characteristics of their distributions, i.e. by means-square deflections ϵ_1 ϵ_2 and ϵ_3 respectively, and the coefficients of mutual correlation \mathbf{r}_{12} , \mathbf{r}_{21} , \mathbf{r}_{23} ,

 r_{13} and r_{31} . These distribution characteristics are found by the following formulae:

$$\sigma_{1}^{2} = \frac{1}{n} \sum_{g=1}^{n} \delta M_{g}^{2}, \quad \sigma_{2}^{2} = \frac{1}{n} \sum_{g=1}^{n} \delta \omega_{g}^{2}, \quad \sigma_{3}^{2} = \frac{1}{n} \sum_{g=1}^{n} \delta \overline{Q}_{g}^{2},$$

$$r_{12} = r_{21} = \frac{\sum_{g=1}^{n} \delta M_{g} \delta \omega_{g}}{n \sigma_{1} \sigma_{2}}; \quad r_{23} = r_{32} = \frac{g-1}{n \sigma_{2} \sigma_{8}};$$

$$r_{13} = r_{31} = \frac{\sum_{g=1}^{n} \delta \overline{Q}_{g} \delta M_{g}}{n \sigma_{3} \sigma_{1}}.$$

$$(10)$$

Then our equations (9) take the form:

$$\begin{vmatrix}
\sigma_{1} \cdot r_{12} + s_{2}^{(1)} \cdot \sigma_{2} + s_{3}^{(1)} \sigma_{3} r_{23} = 0, \\
\sigma_{1} \cdot r_{13} + s_{7}^{(1)} \cdot \sigma_{2} r_{23} + s_{3}^{(1)} \sigma_{3} = 0.
\end{vmatrix}$$
(11)

The solution of eug ions (11), or, what is the same, of equations (10) or (9), is written in the form:

$$s_2^{(1)} = \frac{\sigma_1}{\sigma_2} \cdot \frac{D_{12}}{D_{11}}; \quad s_3^{(1)} = \frac{\sigma_1}{\sigma_8} \cdot \frac{D_{31}}{D_{11}},$$
 (12)

where D_{ik} is the substance of the cofactor:

$$D = \begin{vmatrix} 1 & r_{12} & r_{13} \\ r_{21} & 1 & r_{23} \\ r_{31} & r_{32} & 1 \end{vmatrix}. \tag{13}$$

Similarly resolved is the problem in the case when the coefficients in the interpolative anomalies are taken as unities at ω and $\bar{\delta}$. The following formulae are then obtained:

$$\mu_2 = s_1^{(2)} M + \omega + s_3^{(2)} \overline{\mathbb{Q}},$$
 (14)

$$\mu_3 = s_1^{(3)} M + s_2^{(3)} \omega + \overline{\Omega}, \tag{15}$$

where

$$s_1^{(2)} = \frac{\sigma_2}{\sigma_1} \cdot \frac{D_{12}}{D_{22}}; \quad s_3^{(2)} = \frac{\sigma_2}{\sigma_3} \cdot \frac{D_{23}}{D_{22}},$$
 (16)

$$s_1^{(3)} = \frac{\sigma_8}{\sigma_1} \cdot \frac{D_{31}}{D_{.3}}; \qquad s_2^{(3)} = \frac{\sigma_8}{\sigma_2} \cdot \frac{D_{28}}{D_{31}}.$$
 (17)

As may be easily seen, the three regression equations we obtained are independent, and that is why there will be three interpolative anomalies and not one. These are determined by the formulae:

$$\mu_1 = \overline{M} + s_2^{(1)} \overline{\omega} + s_3^{(1)} \overline{\widehat{Q}},$$
(18)

$$\mu_2 = s_1^{(2)} \overline{M} + \omega + s_2^{(2)} \widehat{Q} , \qquad (19)$$

$$\mu_{3} = S_{1}^{(3)} \overline{M} + S_{2}^{(3)} \overline{\omega} + \overline{\mathcal{Q}}. \tag{20}$$

By their meaning itself, these quantities $s_2^{(1)}$ and $s_3^{(1)}$ are the substance of the regression coefficients of the quantities ω and $\bar{\Omega}$ for -M, thus:

$$-M = \mu_1 + s_2^{(1)} \omega + s_3^{(1)} \overline{Q}$$
 (21)

or

$$-\delta M = s_2^{(1)} \delta \omega + s_3^{(1)} \delta \overline{\mathbb{Q}}. \tag{22}$$

Similarly:

$$-\delta \omega = s_1^{(2)} \delta M + s_3^{(2)} \delta \overline{Q}, \qquad (23)$$

$$-\delta \overline{Q} = s_1^{(3)} \delta M + s_2^{(3)} \delta \omega. \tag{24}$$

Using these formulae one may compute one of our elements by the two others. It is obvious that the so-calculated values of the elements will differ, generally speaking, from those given in the utilized table. The deflections of the computed values from the given ones will be characterized by the mean quadratic deflections Σ_1 , Σ_2 , and Σ_3 respectively for M, ω and $\overline{\lambda}$. It is well known from the theory of linear correlation

for many variables that they may be found in the following manner:

$$\Sigma_1 = \sigma_1 \sqrt{\frac{D}{D_{11}}}, \ \Sigma_2 = \sigma_2 \sqrt{\frac{D}{D_{22}}}, \ \Sigma_3 = \sigma_3 \sqrt{\frac{D}{D_{33}}},$$
 (25)

where D_{11} , D_{22} , D_{33} are cofactors of the elements of determinant's main diagonal (13).

After this it is sufficient to multiply these mean quadratic deflections by the quantity $\sqrt{\frac{n}{n-2}}$ in order to obtain the mean quadratic error per weight unit respectively in the conditional equations:

$$\sqrt{\frac{n}{n-2}} \; \Sigma_1 \; \mathbf{B} \; \delta \mathcal{M} + s_2^{(1)} \delta \boldsymbol{\omega} + s_3^{(1)} \delta \, \overline{\Omega} = 0,$$

$$\sqrt{\frac{n}{n-2}} \; \Sigma_2 \; \mathbf{B} \; s_1^{(2)} \delta \mathcal{M} + \delta \boldsymbol{\omega} + s_3^{(2)} \delta \, \overline{\Omega} = 0,$$

$$\sqrt{\frac{n}{n-2}} \; \Sigma_3 \; \mathbf{B} \; s_1^{(3)} \delta \mathcal{M} + s_2^{(3)} \delta \boldsymbol{\omega} + \delta \, \overline{\Omega} = 0,$$

provided the quantities $s_i^{(k)}$ $(k=1, 2, 3; i=1, 2, 3; i \neq k)$ are considered as those searched for, while ΣM , $\Sigma \omega$ and $\Sigma \bar{\Omega}$ are taken from the Table (6).

Having found the mean square error per unit of weight it is not difficult to compute the mean square errors of the coefficients $s_i^{(k)}$. To do this we still must know the weight of our unknowns. If we transform the formulae of the weights of our unknowns derived from the theory of normal equations, we shall obtain, by introducing in these formulae the distribution characteristics of the quantities M, ω , and $\overline{\Lambda}$:

$$p_{s_i}^{(k)} = n\sigma_i^2 D_{kk}. \tag{26}$$

Then, the mean square errors of the regression coefficients are:

$$\sigma_{s_i^{(k)}} = \sqrt{\frac{n}{n-2}} \frac{\Sigma_k}{\sqrt{p_{s_i^{(k)}}}}$$

$$(i=1, 2, 3; k=1, 2, 3).$$
(27)

It is easy to see that the whole of the just-described method is valid not only for the case of three variables, but also for the case of any number of variables, for in the latter case all formulae from (3) through (27) are easily generalized.

#7. RESULTS OF COMPUTATIONS. CORRELATION COEFFICIENTS

Passing to the review of the concrete results of our computations, we shall begin with the correlation coefficients. The partial correlation coefficient between the quantities M and $\overline{\mathbb{Q}}$ resulted greater for all the ten minor planets, than the two remaining two partial correlation coefficients. The former differs from the unity by two millionths (for Hygiea) in the worst case. This should not be the cause for surprise, for the quantities M and $\overline{\mathbb{Q}}$ vary almost proportionally to the time. As to the remaining partial correlation coefficients between M and ω , ω and $\overline{\mathbb{Q}}$, they are obtained notably different from the unity. The numerical values of the obtained correlation coefficients are given in page 23.

The fact that the particl correlation coefficient \mathbf{r}_{12} was obtained positive, and \mathbf{r}_{23} — negative, shows that the perihelia of all ten minor planets move counterclockwise if one looks from the northern pole of Jupiter's orbit. At the same time for three of the ten planets the

correlation coefficients \mathbf{r}_{12} and \mathbf{r}_{23} resulted smaller than 0.7. This is so because the course of secular variation of ω is distorted by various short-periodical oscillations, and also by the errors in the very determinations of elements of the asteroids' orbit.

Астеронд A Stervid	r ₁₂	r ₂₃	r ₃₁ .
Cercs Pallas Euno Yesta Astrea Hebe Iris Flora Hetis Hygiea	+0.914415 +0.830808 +0.972788 +0.790614 +0.287193 +0.751090 +0.777383 +0.559132 +0.783003 +0.494859	- 0.914470 - 0.830902 - 0.972812 - 0.790837 - 0.287046 - 0.751126 - 0.772215 - 0.559708 - 0.783308 - 0.496041	- 0.999999999999999999999999999999999999

#8. RESULTS OF COMPUTATIONS OF INTERPOLATORY ANOMALIES

By the strength of the indicated causes, and also because of a comparatively small variation of ω , the regression coefficients at ω in the interpolative anomalies μ_1 and μ_2 are unreliably determined, just as are the regression coefficients M and $\overline{\lambda}$ in the expression for μ_2 . These expressions for the interpolative anomalies μ_1 , μ_2 , and μ_3 in the numerical form, are written as follows (the respective mean square errors in the regression coefficients are in parentheses; Σ_i (i=1,2,3) are the errors in the interpolative anomalies):

(In the Table which follows the mean square errors are given at interpolative ammalies. As to the way to compute them, it will be examined in the next paragraph)

MEAN SQUARE ERRORS AT INTERPOLATIVE ANOMALIES

Ne 1. Lepepa (CERES)	ı
$3690 = M + 0.67 (\pm 0.25) \omega + 2.5725 (\pm 0.0003) \overline{\Omega};$	$\Sigma_1 = 0.75$
$14000 = 3.5 \ (\pm 1.0) \ M + \omega + 8.7 \ (\pm 3.0) \ \overline{\Omega};$	$\Sigma_2 = 1.7$
$1536 = 0.38830 \ (\pm \ 0.00005) \ M + 0.26 \ (\pm \ 0.09) \ \bullet + \overline{\Omega};$	$\Sigma_3=0.29.$
№ 2. Паллада (PALLAS)	
$4500 = M + 2.31 (\pm 0.73) \omega + 2.5707 (\pm 0.00025) \overline{\delta};$	$\Sigma_1 = 1.1$
$600 = 0.077 (\pm 0.032) M + \omega + 0.198 (\pm 0.086) \overline{\Omega};$	$\Sigma_2 = 0.27$
$1750 = 0.38899 (\pm 0.00005) M + 0.89 (\pm 0.139) \omega + \overline{\Omega};$	$\Sigma_3 = 0.57$
№ 3. Юнона (EUNO)	
$3820 = M + 1.22 (\pm 0.58) \omega + 2.7184 (\pm 0.00055) \Omega;$	$\Sigma_{\rm j}=1.0$
$120 = 0.031 (\pm 0.015) M + \omega + 0.09 (\pm 0.04) \Omega;$	$\Sigma_2 = 0.16$
$1400 = 0.36785 (\pm 0.00007) M + 0.45 (\pm 0.20) \omega + \Omega;$	$\Sigma_3 = 0.34$
№ 4. Becta (VESTA)	•
$4400 = M + 2.7 (\pm 1.7) \omega + 3.2684 (\pm 0.0012) \Omega;$	$\Sigma_1 = 2.4$
$1100 = 0.24 (\pm 0.15) M + \omega + 0.79 (\pm 0.48) \Omega;$	$\Sigma_2 = 0.70$
$1350 = 0.30596 (\pm 0.00011) M + 0.83 (\pm 0.5) \omega + \Omega;$	$\Sigma_3 = 0.72$
№ 5. Actpes (ASTREA)	•
$370 = M - 0.8 (\pm 0.4) \omega + 2.8660 (\pm 0.00023) \Omega;$	$\Sigma_1 = 0.43$
$250 = -0.14 (\pm 0.07) M + \omega - 0.4 (\pm 0.21) \delta;$	$\Sigma_2 = 0.18$
$130 = 0.34892 (\pm 0.0003) M - 0.29 (\pm 0.14) \omega + \Omega;$	$\Sigma_3 = 0.15$
№ 6. Геба (HEBE)	
$600 = M + 0.6 (\pm 1.5) \omega + 3.1390 (\pm 0.0007) \Omega;$ $270 = 0.03 (\pm 0.07) M + \omega + 0.08 (\pm 0.22) \Omega;$	$\Sigma_1 = 1.6$ $\Sigma_2 = 0.35$
$190 = 0.31857 (\pm 0.00007) M + 0.19 (\pm 0.47) \omega + \overline{0};$	$\Sigma_3 = 0.55$ $\Sigma_3 = 0.51$
	23 — 0.01
NoT. HPHAA. (IRIS)	
$5100 = M - 43.3 (\pm 11.7) \omega + 3.083 (\pm 0.011) \overline{\delta};$	$\Sigma_1 = 30.6$
$15 = -0.0110 (\pm 0.0043) M + \omega - 0.0333 (\pm 0.0042) \overline{\Omega};$	$\Sigma_2 = 0.223$
$1660 = 0.3275 (\pm 0.0036) M - 14.0 (\pm 3.9) \omega + \Omega;$	$\Sigma_3 = 10.0$
№ 8. Флора (Flora)	
$1580 = M + 5.3 (\pm 11) + 3.6324 (\pm 0.007) \Omega;$	$\Sigma_1 = 10.7$
$320 = 0.45 (\pm 0.3) M + \omega + 1.64 (\pm 1.0) \overline{\delta};$	$\Sigma_2 = 3.1$
$436 = 0.2573 (\pm 0.0005) M + 1.46 (\pm 3.0) \omega + \Omega;$	$\Sigma_3 = 2.8$
№ 9. Метида (MET15)	
$200 = M + 19 (\pm 12) \omega + 3.2137 (\pm 0.0015) \Omega;$	$\Sigma_1 = 0.36$
$10 = 0.06 (\pm 0.04) M + \omega + 0.20 (\pm 0.12) \Omega;$	$\Sigma_2 = 0.02$
$62 = 0.31117 (\pm 0.00015) M + 6 (\pm 4) \omega + \overline{\Omega};$	$\Sigma_8 = 0.11$
Nº 10. XHIHER (HYGIEA)	
$1080 = M + 1.43 (\pm 0.8) \omega + 2.1291 (\pm 0.002) \delta;$	$\Sigma_1 = 6.3$
$358 = 0.448 (\pm 0.14) M + \omega + 0.95 (\pm 0.3) \overline{\Omega};$	$\Sigma_2 = 1.9$
$506 = 0.46968 (\pm 0.00023) M + 0.67 (\pm 0.21) \omega + \Omega;$	$\Sigma_3 = 1.6$

#9. COMPUTATION OF DISPERSIONS OF INTERPOLATIVE ANOMALIES

The dispersion of the interpolative anomaly ² is obviuosly equal to the quantity:

$$\sigma_{\overline{g}}^{2} = \frac{1}{n} \sum_{g=1}^{n} \left[\mu - (s_{1} M_{g} + s_{2} \omega_{g} + s_{3} \overline{Q}_{g}) \right]^{2}.$$
 (28)

Introducing intro

$$\sigma_{\mu}^{2} = \frac{1}{n} \sum_{g=1}^{n} (s_{1} \delta M_{g} + s_{2} \delta \omega_{g} + s_{3} \delta \overline{\Omega}_{g})^{2},$$

$$\mu = s_{1} \overline{M} + s_{2} \widetilde{\omega} + s_{3} \overline{\widetilde{\Omega}},$$
(29)

or

Thus, the dispersions of interpolative anomalies μ_1 , μ_2 and μ_3 are equal respectively to the quantities Σ_1 , Σ_2 and Σ_3 .

#10. DISCUSSION OF THE RESULTS OF COMPUTATION OF INTERPOLATIVE ANOMALIES

The above-presented material provides the possibility of reaching a series of conclusions.

Let us examine in the first place the question of the comparative quality of the obtained interpolative anomalies for each of the considered minor planets.

It is most natural to take as the criterion of this comparative quality the degree of satisfactory fulfillment of the "near-constancy" requirement for the interpolative anomaly within the examined time interval. As to its quantitative characteristic, one may take the degree of

closeness to the unity of the general correlation coefficient. If we designate by R_1 , R_2 and R_3 the respective general correlation coefficients for rhe interpolative anomalies μ_1 , μ_2 and μ_3 , we may derive formulae expressing them through partial correlation coefficients:

$$R_{i} = \sqrt{1 - \frac{D}{D_{iii}}}; \quad i = 1, 2, 3.$$
 (30)

as this is being done in the theory of linear correlation for many variables (see for example the book by Romanovskiy [3]).

It is clear from these formulae, that the greater $D_{i\dot{i}}$ the greater will be the corresponding total correlation coefficient R_i . But

$$D_{11} \stackrel{\mathbf{T}}{=} 1 - r_{23}^2; \ D_{22} = 1 - r_{31}^2; \ D_{33} = 1 - r_{12}^2. \tag{31}$$

We thus may already estimate the quality of interpolative anomalies by the partial correlation coefficients themselves. Looking over the Table of these coefficients, given in #7, we see that the correlation coefficient \mathbf{r}_{31} exceeds the remaining \mathbf{r}_{12} and \mathbf{r}_{23} for all the ten minor planets. We conclude on the basis of formulae (30) and (31) that the total correlation coefficient \mathbf{R}_2 must be smaller than the remaining \mathbf{R}_1 and \mathbf{R}_3 . This leads us to conclude that the first and the thirs interpolative anomalies are better for the first ten asteroids examined by us, than the second interpolative anomaly.

Further, the attention is called to the fact that the mean-square ærrors of the coefficients at ω exceed the corresponding mean-square errors for the coefficients at M and $\overline{\mathfrak{A}}$.

Just as the preceding one, this fact is explained by the irregularity of the motion of the perihelion and by errors in the determination of the quantity ω .

The closeness of the coefficients at $\overline{\mathcal{N}}$ in the first anomaly, and near \mathbf{M} in the third anomaly to the direct and inverse ratios of the mean motions of Jupiter and of the asteroid is also explained by the same fact. Besides, through more elaborate study of the results obtained one may see, that within the limits of coefficient errors, the latter are proportional to one another in the first and third interpolative anomalies, just as are the interpolative anomalies themselves. This too is explained by the above-described cause.

In order to obtain a more complete representation on the quality of the interpolative anomalies constructed by us, it is interesting to compare them with those anomalies, which were utilized for minor planets under the denomination of "Delaunay anomalies".

#11. COMPARISON OF INTERPOLATIVE ANOMALIES WITH THE DELAUNAY ANOMALIES

 $W_{\mathbf{e}}$ shall designate as the asculating Delaunay anomaly, or simple "Delaunay anomaly", the quantity :

$$\Delta = q_1 M + q_2 \overline{\mathbb{Q}}, \tag{32}$$

where the coefficients \mathbf{q}_2 and \mathbf{q}_1 are related among themselves as the mean motions of the asteroid and of J_{UD} iter, while the mean motion of the former is taken for a specific osculation epoch. If we take for that osculation epoch the initial osculation epoch from Table 1,

we shall obtain:

Ceres	n : n3	2.572987
Euno	n _j : n	0.367661
Astrea	nj: n	0.348462
Pallas	n": nj	0.57277
Vesta	n: nj	3.26914
Hebe	nj: n	0.318559
H ygiea	nj: n	0.469701

Because of the absence of a sufficient number of systems of osculating elements, such calculations have not been made for Flora and Metis. For Iris the mean motion for that epoch is /-very unreliably.

To compare the interpolative anomaly with the Delaunay anomaly, we shall examine the dispersions of both these anomalies. The dispersion of the latter is equal to that of the quantity:

or
$$\epsilon_{D} = q_{1}\delta M + q_{2}\delta\overline{Q}, \qquad (33)$$
and
$$\Delta = q_{1}M + q_{2}\overline{Q} \qquad (34)$$

$$\sigma_{D}^{2} = \frac{1}{n} \sum_{g=1}^{n} [\Delta - (q_{1}M_{g} + q_{2}\overline{Q}_{g})]^{2}, \qquad (35)$$
or
$$\sigma_{D}^{2} = \frac{1}{n} \sum_{g=1}^{n} (q_{1}\delta M_{g} + q_{2}\delta\overline{Q})^{2}. \qquad (36)$$

It is obvious that the dispersion of the interpolative anomaly $\mu=s_1M+s_2\omega+s_3N$ is equal to the dispersion of the quantity:

$$\varepsilon_{\mu} = s_1 \hat{o} M + s_2 \hat{o} \omega + s_3 \hat{o} \overline{O}, \tag{37}$$

as we already had the opportunity to see it in #9. The only thing that remains is the comparison of both quantities (33) and (37).

It is most appropriate to plot these values in a graph. All such graphs are presented at the end of the paper. It is obvious that only the first and the third interpolative anomalies may be compared with the Delaunay anomaly. But for a better proportionality of the latter, providing a greater facility in graph construction for certain planets, the Delaunay and the interpolation anomalies were compared in the form:

$$\Delta_{1} = M + \frac{n}{n_{j}} \Omega,$$

$$\mu_{1} = M + s_{2}^{(1)} \omega + s_{3}^{(1)} \overline{\Omega}.$$
(38)

But for other minor planets it was done in the form:

$$\Delta^2 = \frac{n_j}{n} M + \overline{Q}, \tag{39}$$

$$\mu_3 = s_1^{(3)} M + s_2^{(3)} \omega + \bar{Q}. \tag{15}$$

These graphs clearly show that the interpolative anomalies are more constant quantities, than the Delaunay anomalies.

In conclusion, the author wishes to express his gratitude to his scientific guide — Professor N.D.Moiseyev, who provided him with valuable indications in the course of the completion of the present work, and who also contributed that valuable material on the systems of osculating elements for minor planets, from which the above-expounded results have sprung.

****** THE END ******

Translated by ANDRE L. BRICHANT

for the

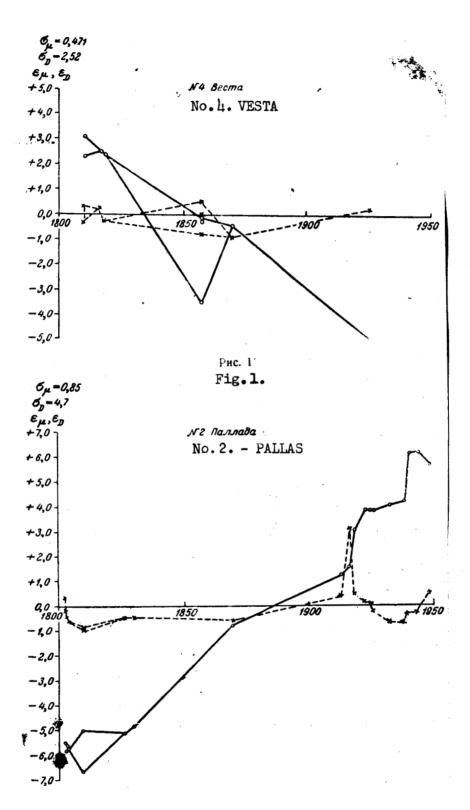
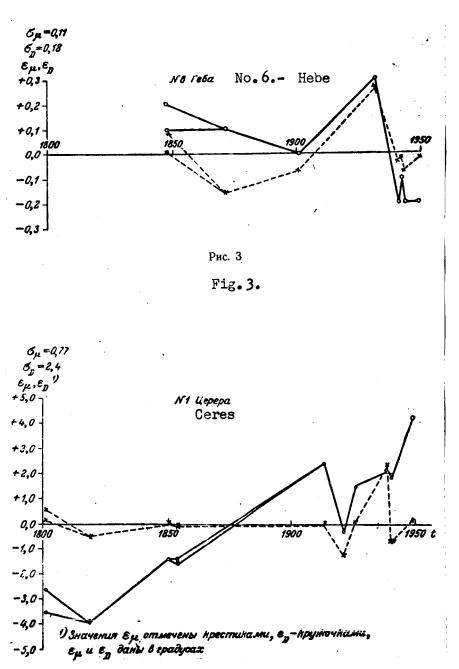


Fig. 2.



1) The values ϵ_μ are indicated by crosses, ϵ_D — by small circles and ϵ_μ and ϵ_D are given in degrees.

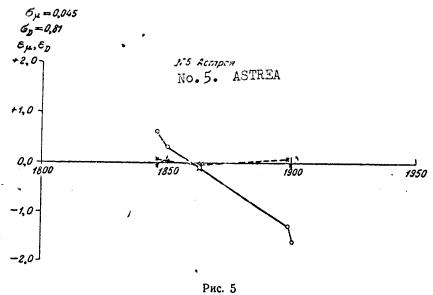


Fig.5

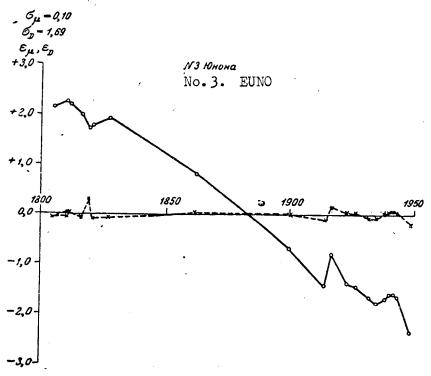


Fig. 6

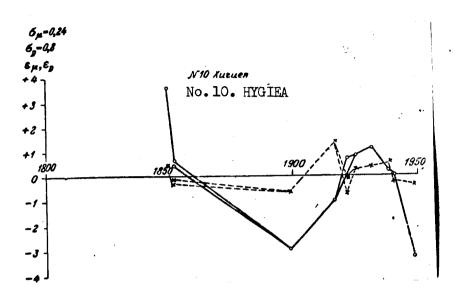


Fig. 7.

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